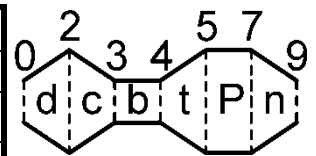


	2	3	4	5	7	9	
	diff	comp	CC	Tur	AB	No	
c_p	1004			1152		1243	J/kgK
γ	1.4			1.33		1.3	
π	0.96	10	0.95		0.98	0.97	
$\eta, e_{c,t}$		0.9	0.99	0.9	0.99		
Tt				1750		2250	
M0	2		po/p9	1	QR	42800	kJ/kgK
T0	250	K	p0	101,300	Pa	η_m	0.99
k	0.28571			0.24812		0.23077	
R	286.857			285.835		286.846	kJ/kgK
a0	316.86	m/s	V0	633.719	m/s		
$\tau_r = \psi$	1.8						
Section	c	t	λ				t
τ	2.07711	0.76251	8.03187			π	0.29696
Section	0	2	3	4	5	7	9
Tt	450	450	934.701	1750	1334.4		1334.4
pt	792,617	760,912	7.61E+06	7.23E+06	2.15E+06		2.08E+06
Section	f	Pt9/p9	M9	Tt9/T9	T9/T0	a9	
3	0.0267	2.06E+01	2.60209	2.11719	2.52107	489.492	m/s
5	0	V9		V9/a0	V9/a0 eff	F/ma0	Num
		1273.7	m/s	4.01977	4.01977	2.1271	1.26E+06
Section	t	th	p	0	TSFC*1e3		
η	0.913	0.55303	0.67581	0.37374	0.039617		



$$k = \frac{\gamma - 1}{\gamma} = \frac{0.4}{1.4} = 0.2867 \quad k_5 = \frac{\gamma_5 - 1}{\gamma_5} = \frac{0.33}{1.33} = 0.2481$$

$$k_9 = \frac{\gamma_9 - 1}{\gamma_9} = \frac{0.3}{1.3} = 0.2308 \quad R_9 = 0.2308 \cdot 1243 = 286.8 \text{ J/kgK}$$

$$R = k c_p = 0.2867 \cdot 1004 = 286.9 \text{ J/kgK} \quad R_5 = 0.2481 \cdot 1152 = 285.8 \text{ J/kgK}$$

$$a_0 = \sqrt{\gamma R T_0} = 1.4 \cdot 286.9 \cdot 250 = 316.9 \text{ m/s}$$

$$V_0 = M_0 \cdot a_0 = 2.00 \cdot 316.9 = 633.7 \text{ m/s}$$

Presa d'aria

$$\tau_r = \psi_0 = 1 + \frac{\gamma - 1}{2} M_0^2 = 1 + 0.2 \cdot 4 = 1.800$$

$$T_{t0} = T_{t2} = T_0 \tau_r = 250 \cdot 1.800 = 450 \text{ K}$$

$$p_{t0} = \tau_r^{\frac{1}{k}} p_0 = 1.800^{\frac{1}{0.2867}} 1.013 \cdot 10^5 = 792.6 \text{ kPa}$$

$$p_{t2} = p_{t0} \pi_d = 792.6 \cdot 0.96 = 760.9 \text{ kPa}$$

Compressore

$$\tau_c = \pi_c^{\frac{k}{\gamma_c}} = 10^{\frac{0.2867}{0.90}} = 2.077$$

$$p_{t3} = p_{t2} \pi_c = 760.9 \cdot 10 = 7609 \text{ kPa}$$

$$T_{t3} = T_{t2} \tau_c = 450 \cdot 2.077 = 934.7 \text{ K}$$

Camera di combustione

$$p_{t4} = p_{t3} \pi_b = 7609 \cdot 0.95 = 7229 \text{ kPa}$$

$$\tau_\lambda = \frac{c_{p5} T_{t4}}{c_p T_0} = \frac{1152 \cdot 1750}{250 \cdot 1004} = 8.032$$

$$f = \frac{\tau_\lambda - \tau_c \tau_r}{Q_R \eta_b / (c_p T_0) - \tau_\lambda} = \frac{8.032 - 2.077 \cdot 1.8}{42.8 \cdot 10^6 \cdot 0.99 / (250 \cdot 1004) - 8.032} = 0.02670$$

Turbina

$$\tau_t = 1 - \frac{\tau_r (\tau_c - 1)}{\eta_m (1 + f) \tau_\lambda} = 1 - \frac{1.8 \cdot 1.077}{0.99 \cdot 1.027 \cdot 8.032} = 0.7625$$

$$\pi_t = \tau_t^{\frac{1}{k_5 \gamma_t}} = 0.7625^{\frac{1}{0.2481 \cdot 0.900}} = 0.2970$$

$$p_{t5} = p_{t4} \pi_t = 7229 \cdot 0.2970 = 2147 \text{ kPa}$$

$$T_{t5} = T_{t4} \tau_t = 1750 \cdot 0.7625 = 1334 \text{ K}$$

Ugello (AB Off)

$$p_{t9} = p_{t5} \pi_n = 2147 \cdot 0.97 = 2082 \text{ kPa}$$

$$\frac{p_{t9}}{p_9} = \frac{p_{t9} p_9}{p_9 p_0} = \frac{2082}{101.3} \cdot 1 = 20.55$$

$$\psi_9 = \left(\frac{p_{t9}}{p_9} \right)^{k_5} = 20.55^{0.2481} = 2.117$$

$$M_9 = \sqrt{\frac{2}{\gamma_5 - 1} [\psi_9 - 1]} = \sqrt{\frac{2}{.33} [1.117]} = 2.602$$

$$M_9 > 1 \rightarrow p_9 = p_0 \text{ (Si dovrebbe verificare rapporto aree)}$$

$$T_9 = T_{t9} / \psi_9 = T_{t5} / \psi_9 = 1334 / 2.117 = 630.3 \text{ K}$$

$$a_9 = \sqrt{\gamma_5 R_5 T_9} = \sqrt{1.33 \cdot 285.8 \cdot 630.3} = 489.5 \text{ m/s}$$

$$V_9 = M_9 \cdot a_9 = 2.602 \cdot 489.5 = 1274 \text{ m/s}$$

$$\frac{V_9}{a_0} = \frac{1274}{316.9} = 4.020$$

Spinta

$$V_{9,e} = V_9 \left[1 + \frac{1 - \frac{p_0}{p_9}}{\gamma_9 M_9^2} \right] = V_9$$

$$\frac{F_u}{\dot{m}_0 a_0} = (1 + f) \frac{V_{9,e}}{a_0} - M_0 = 1.027 \cdot 4.020 - 2.00 = 2.127$$

$$TSFC = \frac{f}{F_u/\dot{m}_0} = \frac{0.02670 \cdot 10^3}{2.127 \cdot 316.9} = 0.03962 \frac{g}{s N}$$

$$\eta_{th} = \frac{a_0^2 [(1 + f) V_{9,e}^2 / a_0^2 - M_0^2]}{2f Q_R} =$$

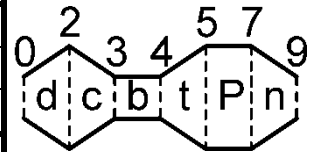
$$\eta_{th} = \frac{316.9^2 [1.027 \cdot 4.020^2 - 2.000^2]}{2 \cdot 0.02670 \cdot 42.8 \cdot 10^6} = \frac{1.264 \cdot 10^6}{2.286 \cdot 10^6} = 0.553 = 55.3\%$$

$$\eta_p = \frac{2 \frac{F_u}{\dot{m}_0 a_0} a_0 V_0}{a_0^2 [(1 + f) V_{9,e}^2 / a_0^2 - M_0^2]} = \frac{2 \cdot 2.127 \cdot 316.9 \cdot 633.7}{1.264 \cdot 10^6} = 0.675 = 67.5\%$$

$$\eta_0 = \eta_{th} \eta_p = 0.553 \cdot 0.675 = 0.374 = 37.4\%$$

AB ON

Section	0	2	3	4	5	7	9
Tt	450	450	934.701	1750	1334.4		1334.4
pt	792,617	760,912	7.61E+06	7.23E+06	2.15E+06		2.08E+06
Section	f	Pt9/p9	M9	Tt9/T9	T9/T0	a9	
3	0.0267	2.06E+01	2.60209	2.11719	2.52107	489.492	m/s
5	0	V9		V9/a0	V9/a0 eff	F/ma0	Num
		1273.7	m/s	4.01977	4.01977	2.1271	1.26E+06
Section	t	th	p	0	TSFC*1e3		
η	0.913	0.55303	0.67581	0.37374	0.039617		



$$k_9 = \frac{\gamma_9 - 1}{\gamma_9} = \frac{0.3}{1.3} = 0.2308 \quad R_9 = 0.2308 \cdot 1243 = 286.8 J/kgK$$

Post Bruciatore

$$p_{t7} = p_{t5} \pi_{AB} = 2147 \cdot 0.98 = 2104 kPa$$

$$\tau_{\lambda,AB} = \frac{c_{p9} T_{t7}}{c_p T_0} = \frac{1243 \cdot 2250}{250 \cdot 1004} = 11.14$$

$$f_{AB} = \frac{(1 + f)(\tau_{\lambda,AB} - \tau_{\lambda} \tau_t)}{Q_{R,AB} \eta_{b,AB} / (c_p T_0) - \tau_{\lambda,AB}} = \frac{1.027(11.14 - 8.032 \cdot 0.7625)}{\frac{42.8 \cdot 10^6 \cdot 0.99}{250 \cdot 1004} - 11.14} = 0.03268$$

Ugello

$$p_{t9} = p_{t7}\pi_n = 2104 \cdot 0.97 = 2041 \text{ kPa}$$

$$\frac{p_{t9}}{p_9} = \frac{p_{t9}}{p_9} \frac{p_9}{p_0} = \frac{2041}{101.3} \cdot 1 = 20.14$$

$$\psi_9 = \left(\frac{p_{t9}}{p_9} \right)^{k_5} = 20.14^{0.2308} = 2.000$$

$$M_9 = \sqrt{\frac{2}{\gamma_9 - 1} [\psi_9 - 1]} = \sqrt{\frac{2}{.30} [1.000]} = 2.582$$

$$M_9 > 1 \rightarrow p_9 = p_0 \text{ (Si dovrebbe verificare rapporto aree)}$$

$$T_9 = T_{t9}/\psi_9 = 2250/2.000 = 1125 \text{ K}$$

$$a_9 = \sqrt{\gamma_9 R_9 T_9} = \sqrt{1.30 \cdot 286.8 \cdot 1125} = 647.6 \text{ m/s}$$

$$V_9 = M_9 \cdot a_9 = 2.582 \cdot 647.6 = 1672 \text{ m/s}$$

$$\frac{V_9}{a_0} = \frac{1672}{316.9} = 5.277$$

Spinta

$$V_{9,e} = V_9 \left[1 + \frac{1 - \frac{p_0}{p_9}}{\gamma_9 M_9^2} \right] = V_9$$

$$\frac{F_u}{\dot{m}_0 a_0} = (1 + f + f_{AB}) \frac{V_{9,e}}{a_0} - M_0 = (1 + 0.0267 + 0.0327) \cdot 5.277 - 2.00 = 3.591$$

$$TSFC = \frac{f + f_{AB}}{F_u/\dot{m}_0} = \frac{0.05938 \cdot 10^3}{3.591 \cdot 316.9} = 0.05219 \frac{\text{g}}{\text{s N}}$$

$$\eta_{th} = \frac{a_0^2 [(1 + f + f_{AB}) V_{9,e}^2 / a_0^2 - M_0^2]}{2(f + f_{AB}) Q_R} =$$

$$\eta_{th} = \frac{316.9^2 [1.059 \cdot 5.277^2 - 2.000^2]}{2 \cdot 0.05938 \cdot 42.8 \cdot 10^6} = \frac{2.561 \cdot 10^6}{5.083 \cdot 10^6} = 0.504 = 50.4\%$$

$$\eta_p = \frac{2 \frac{F_u}{\dot{m}_0 a_0} a_0 V_0}{a_0^2 [(1 + f + f_{AB}) V_{9,e}^2 / a_0^2 - M_0^2]} = \frac{2 \cdot 2.127 \cdot 316.9 \cdot 633.7}{1.264 \cdot 10^6} = 0.563 = 56.3\%$$

$$\eta_0 = \eta_{th} \eta_p = 0.504 \cdot 0.563 = 0.284 = 28.4\%$$